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(54) **UNMANNED AIR VEHICLE WEAPON ADAPTER**

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See application file for complete search history.

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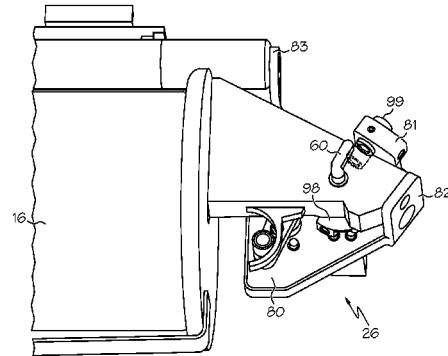
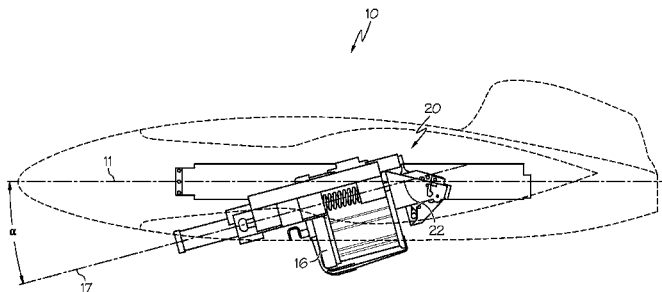
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(57) **ABSTRACT**

In some embodiments, an adapter for attaching a handheld firearm to an unmanned air vehicle comprises a mounting assembly configured for attachment to said firearm and to the unmanned air vehicle, and an actuation assembly comprising an actuation device configured to actuate a trigger of the handheld firearm.

**21 Claims, 13 Drawing Sheets**



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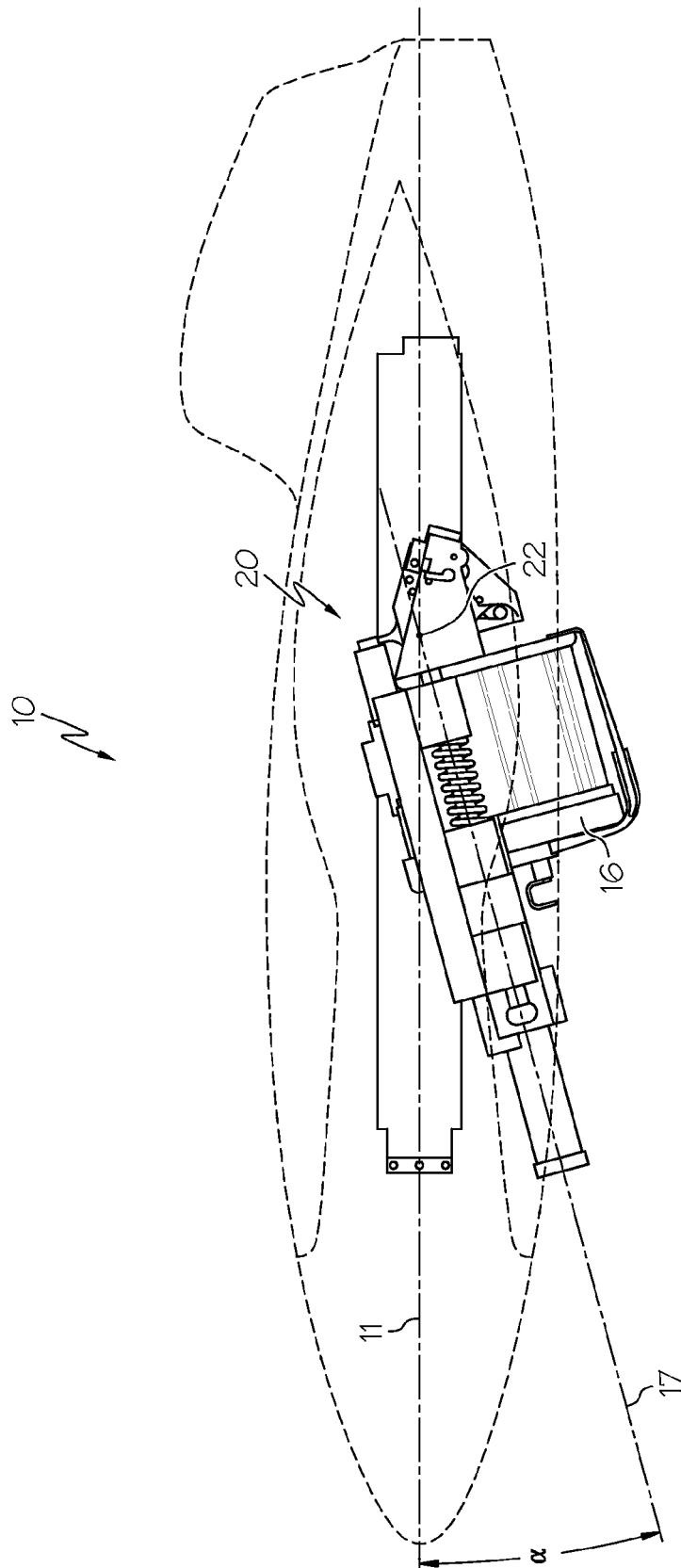


FIG. 1

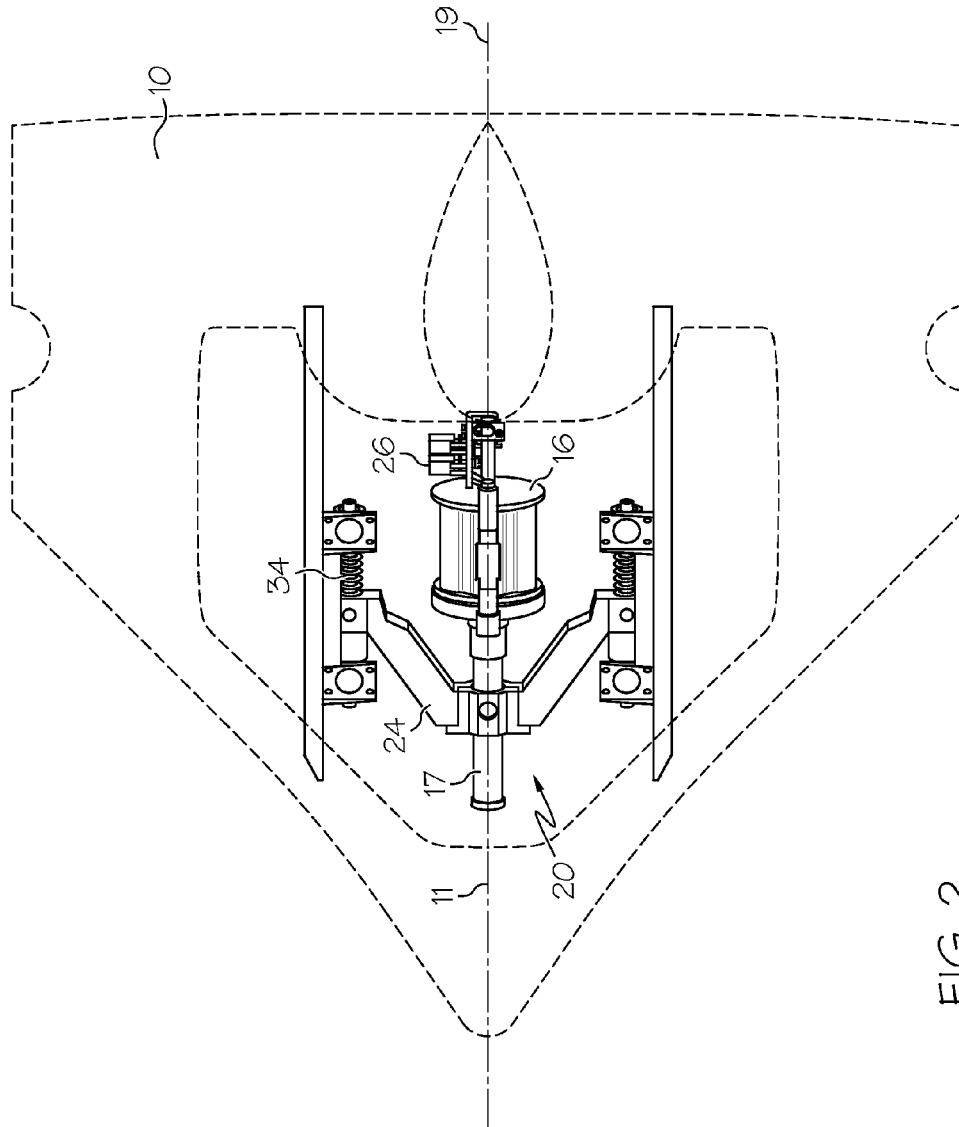


FIG. 2

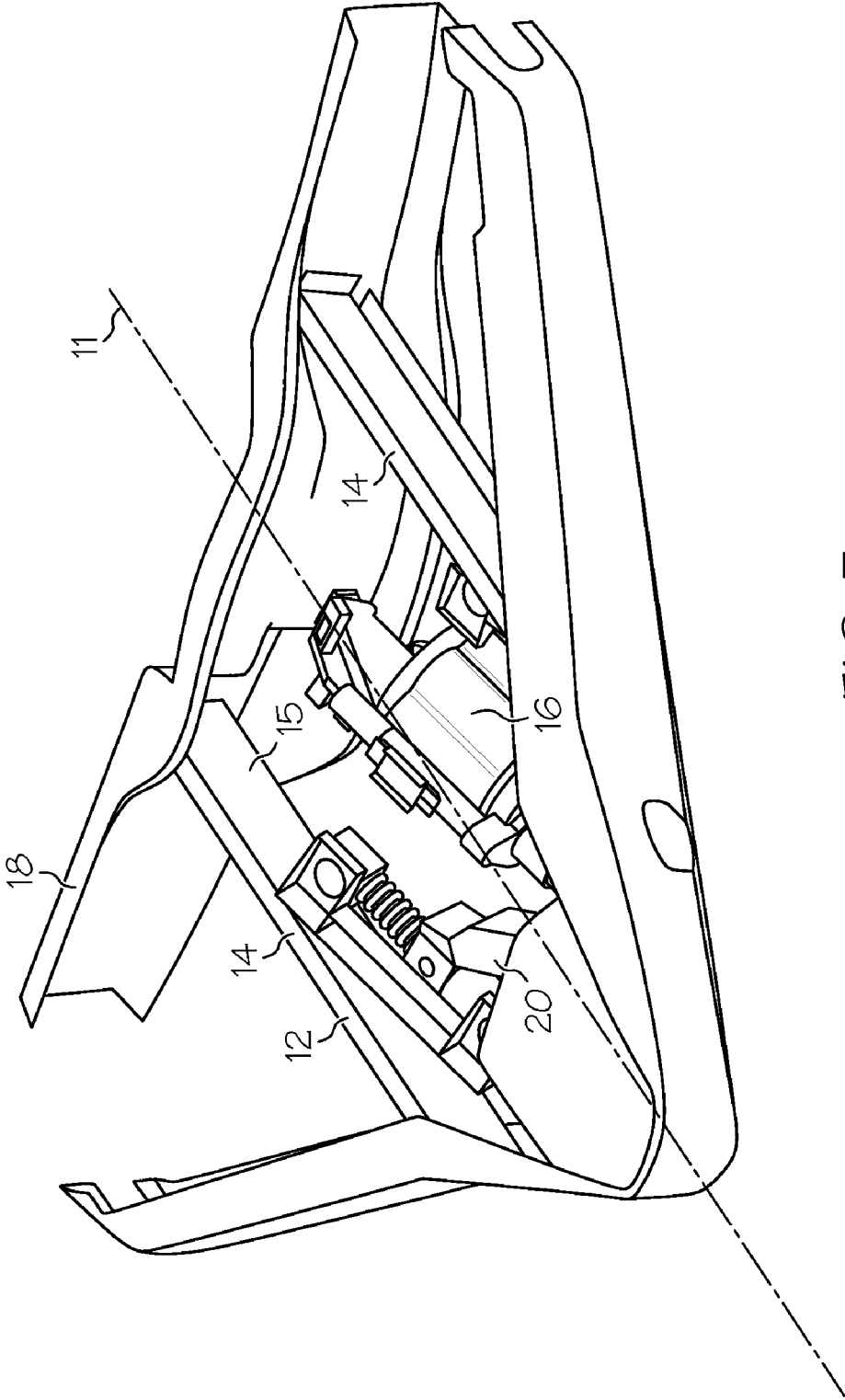


FIG. 3

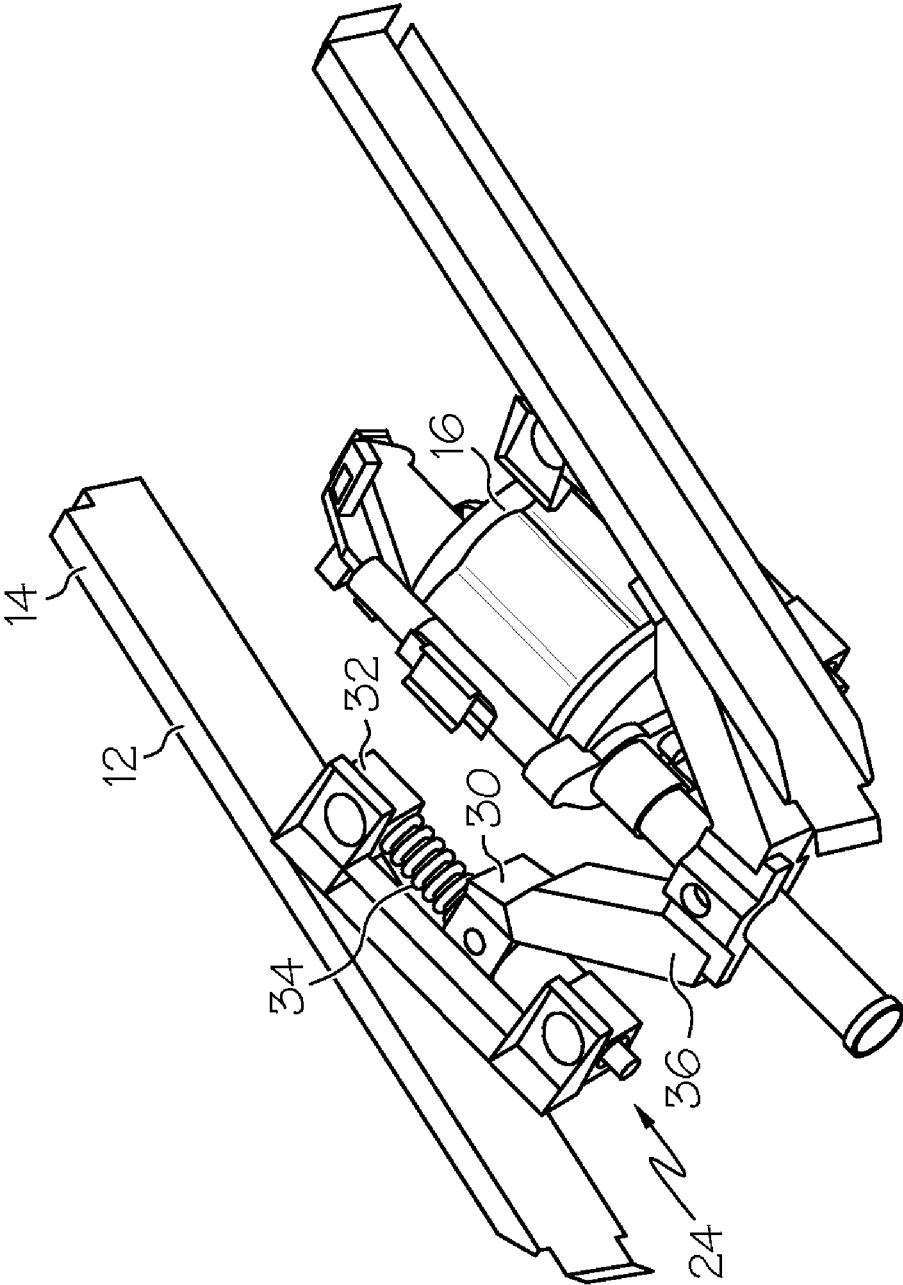


FIG. 4

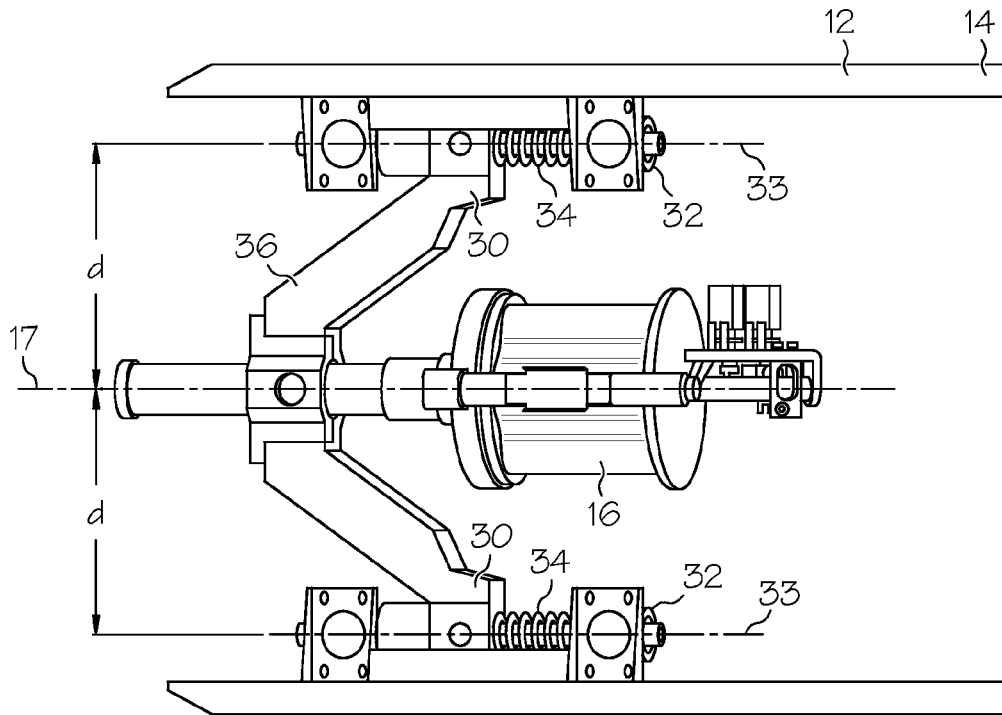


FIG. 5

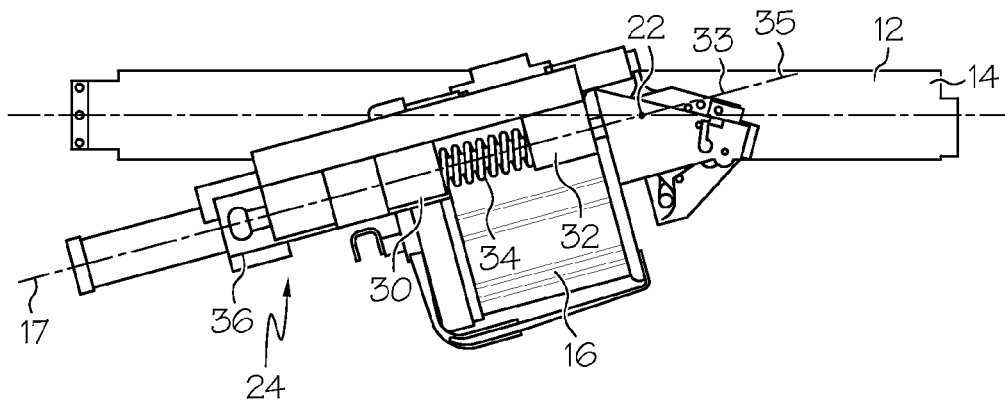


FIG. 6

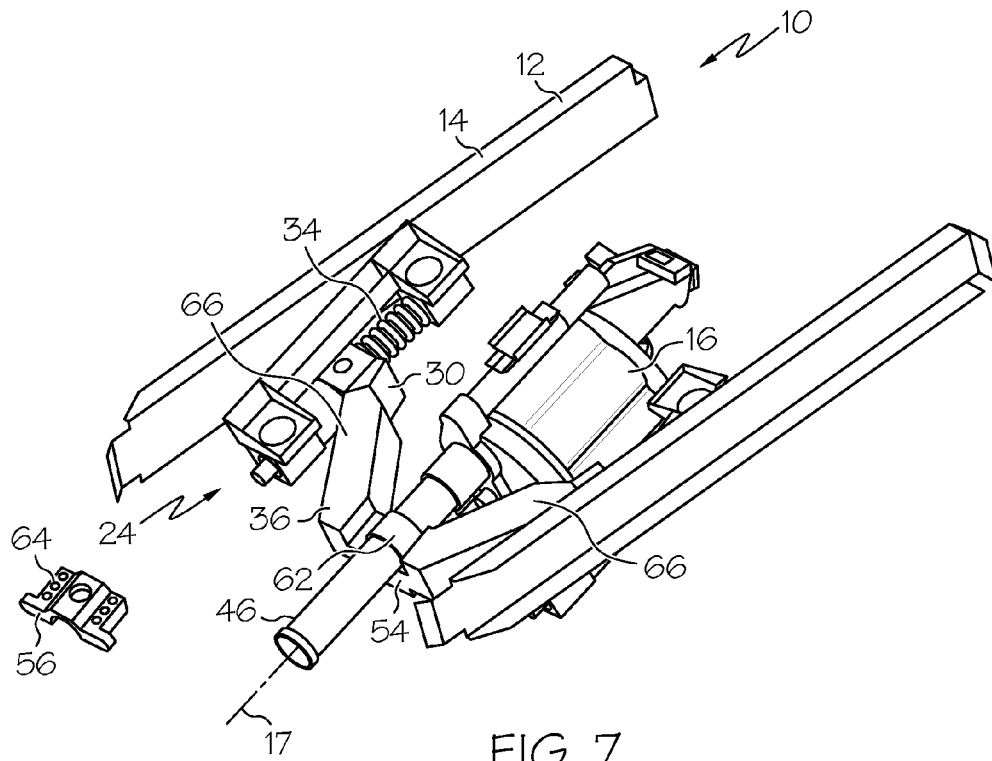


FIG. 7

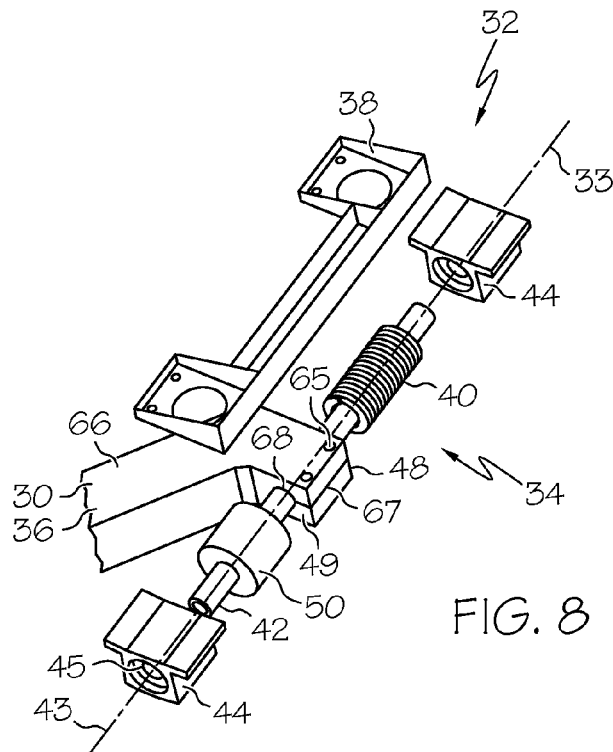
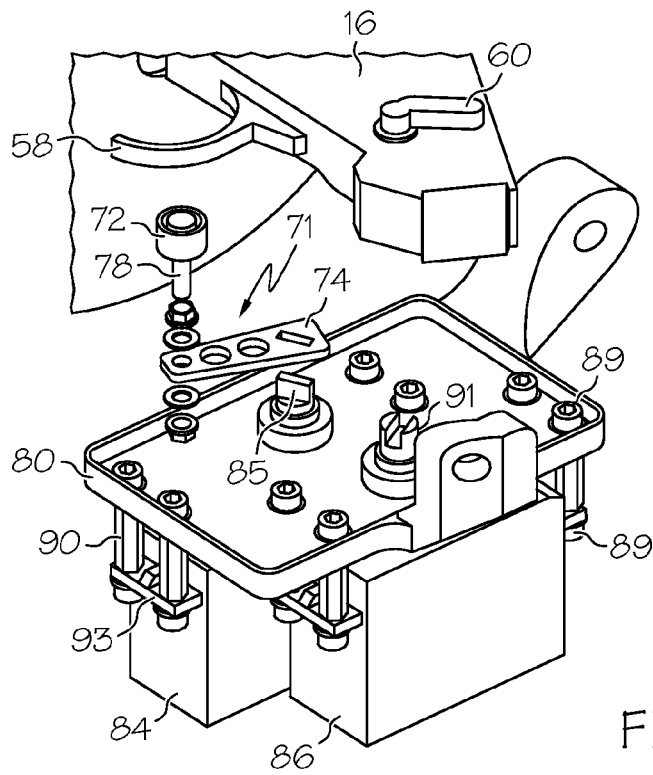
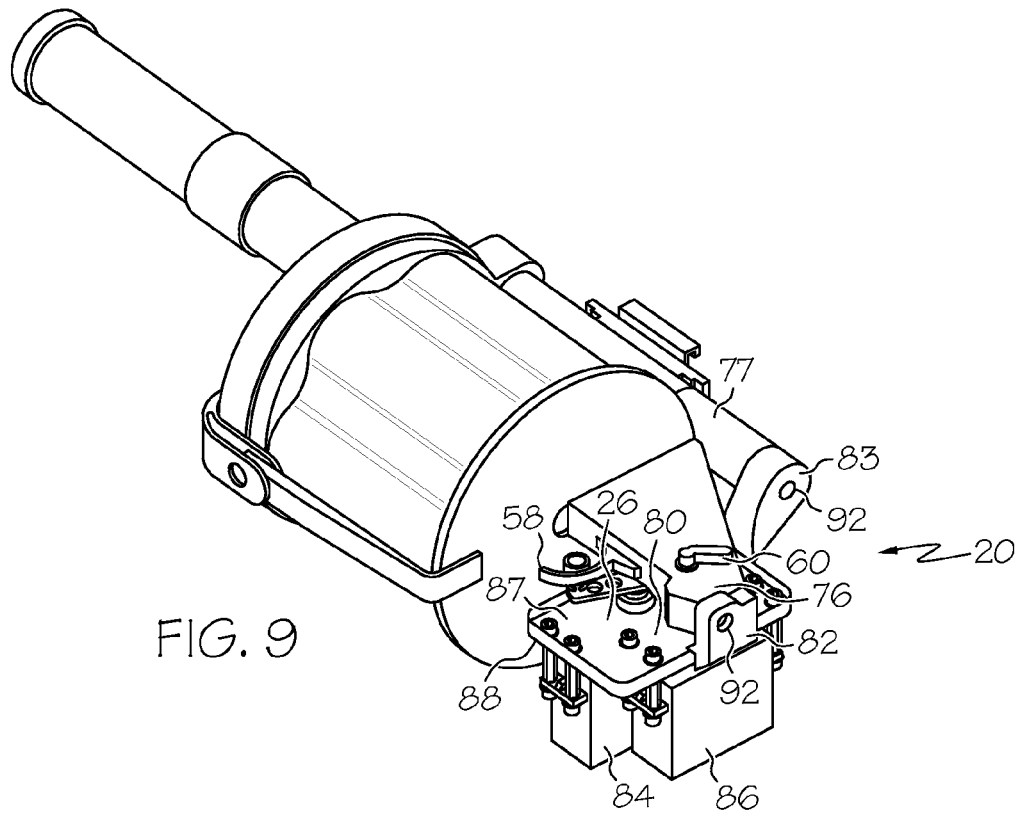
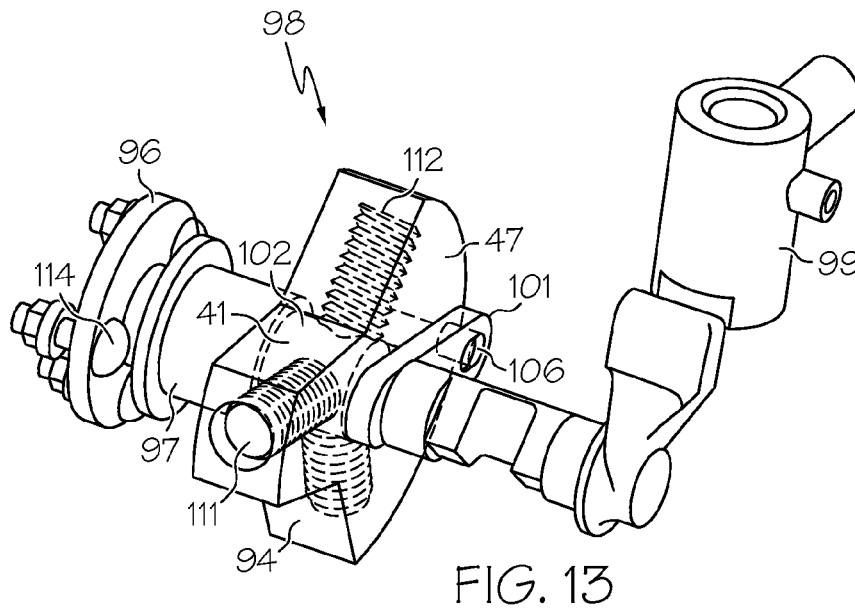
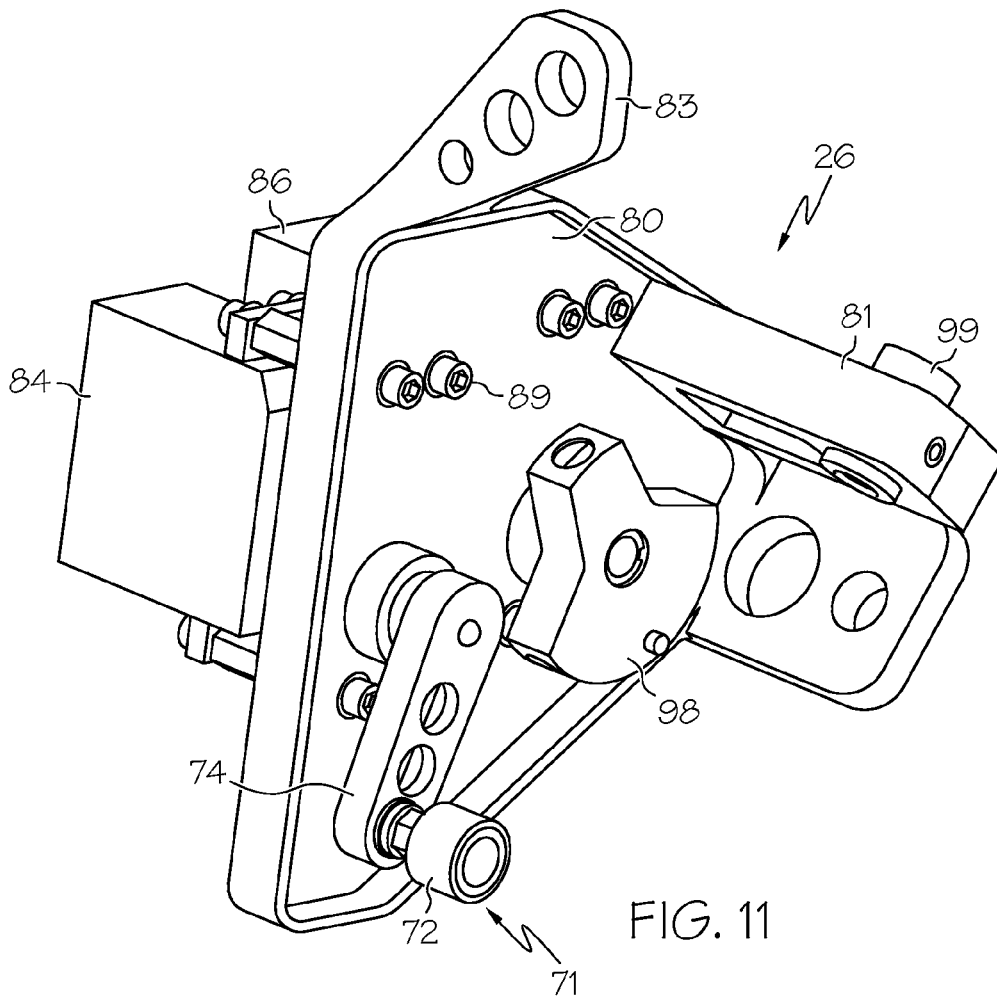


FIG. 8







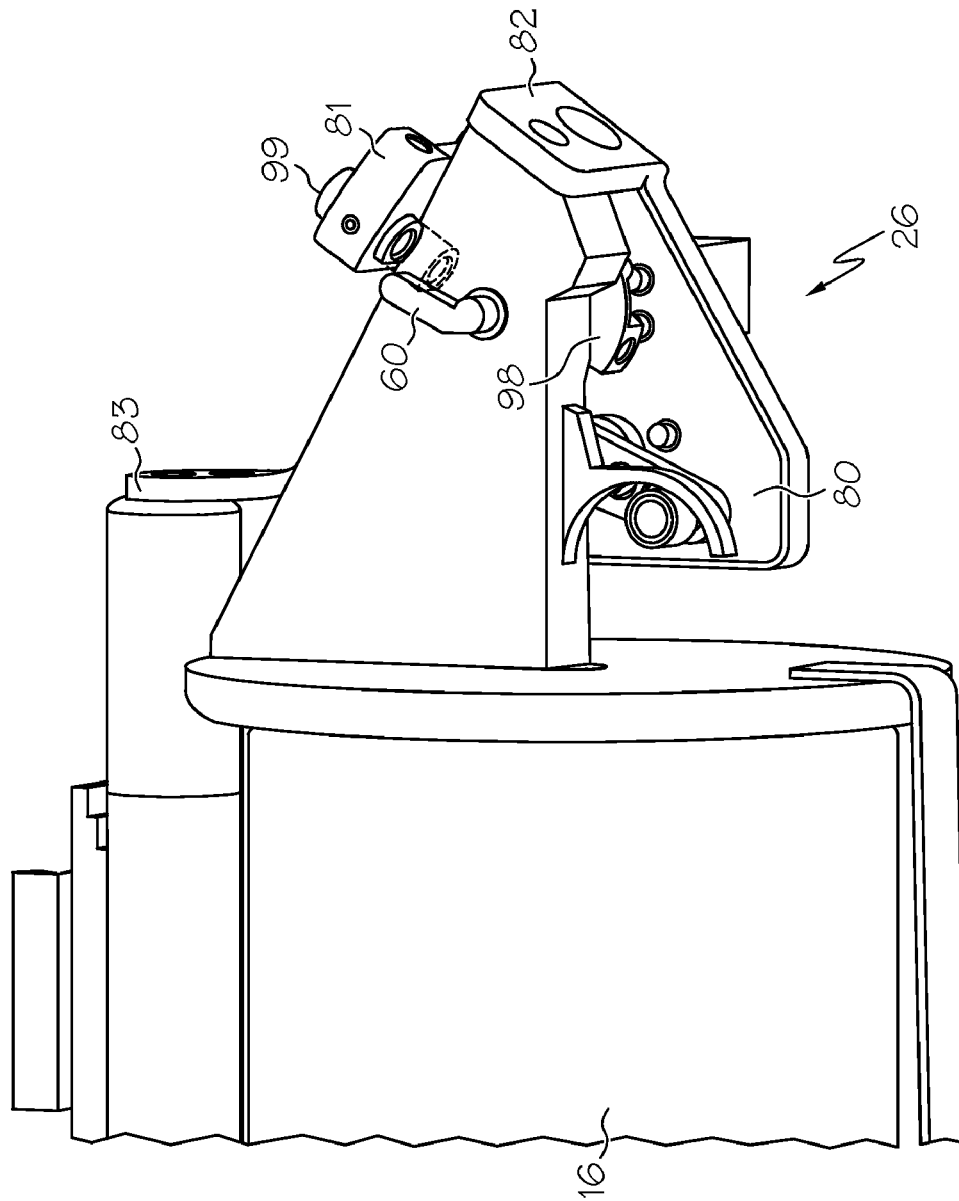


FIG. 12

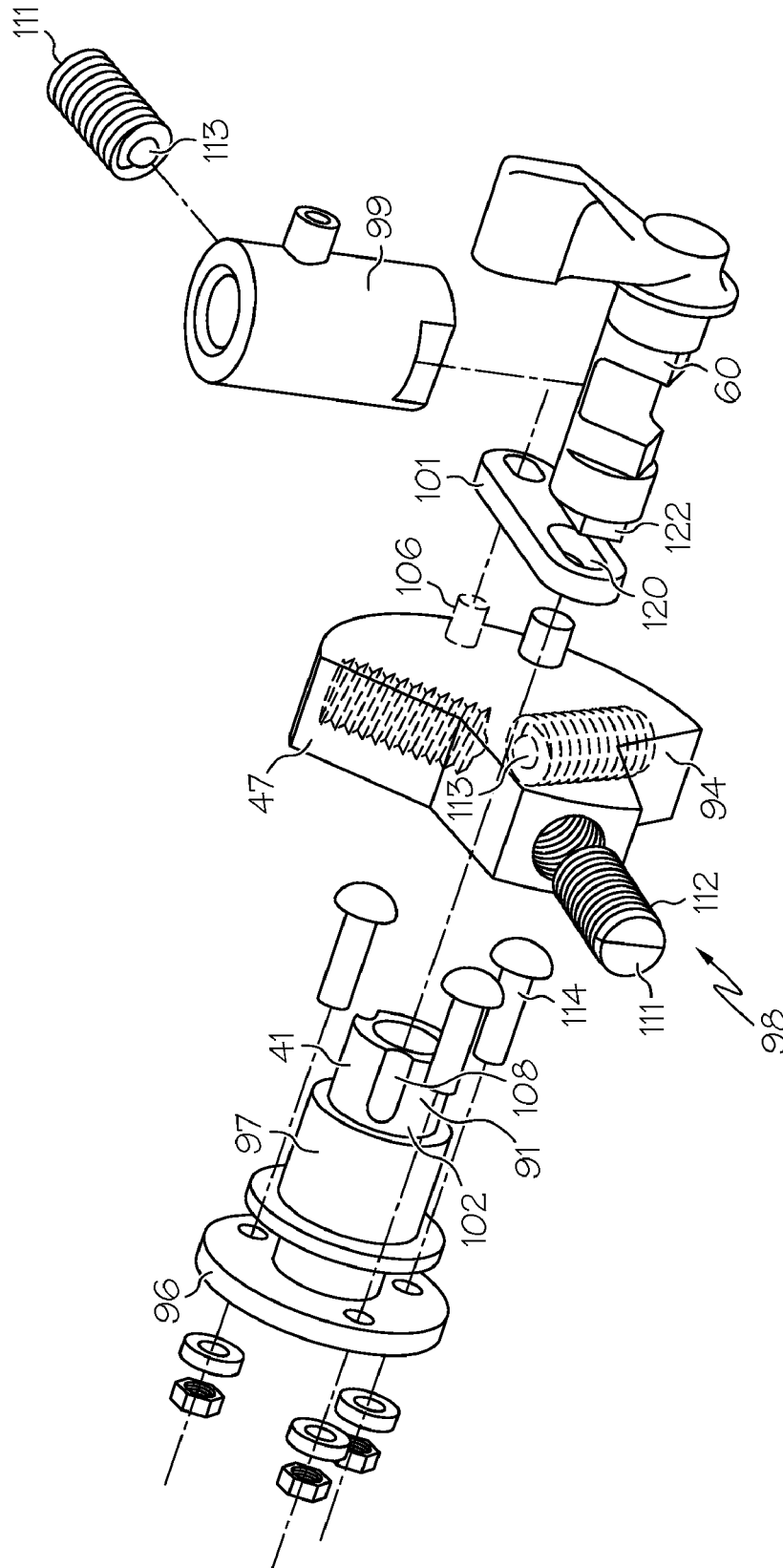


FIG. 14

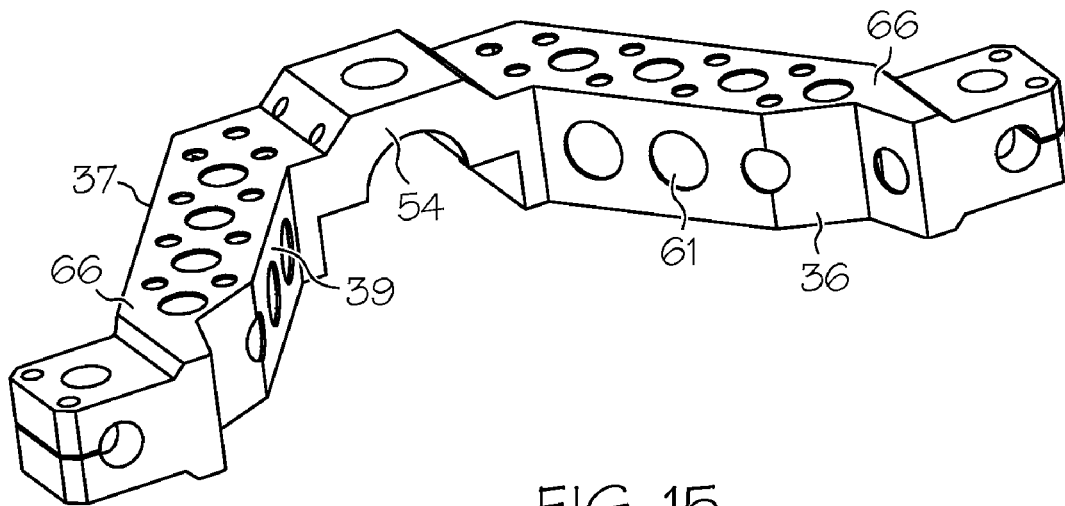


FIG. 15

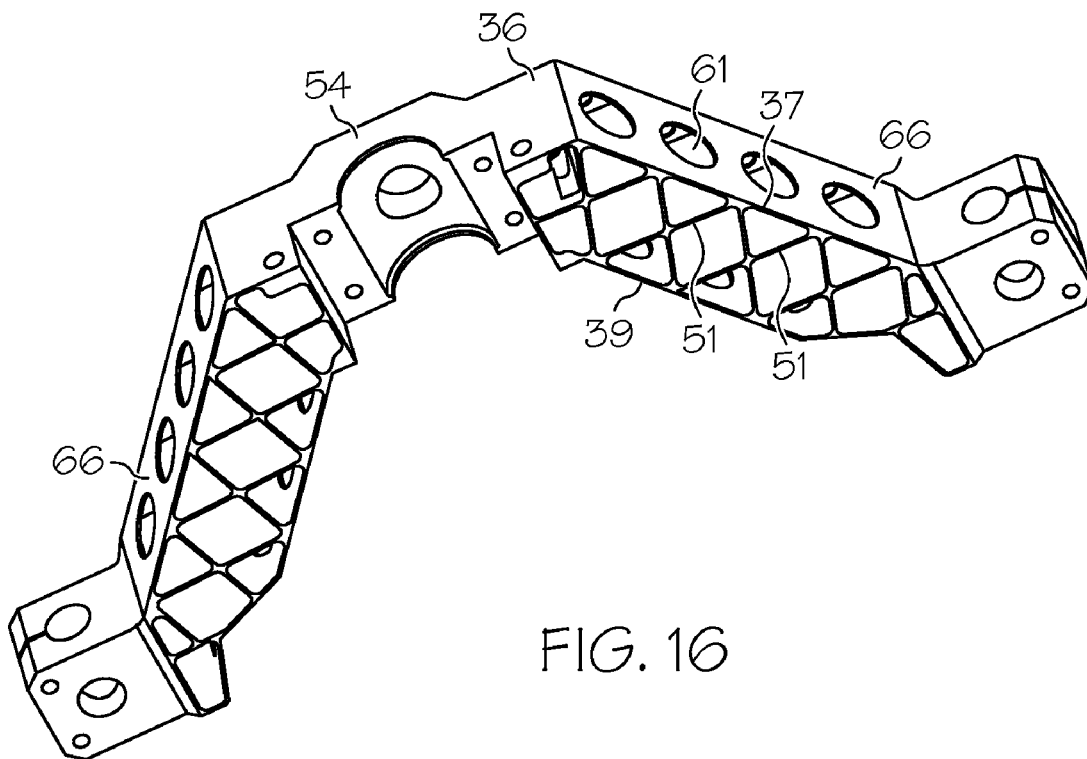


FIG. 16

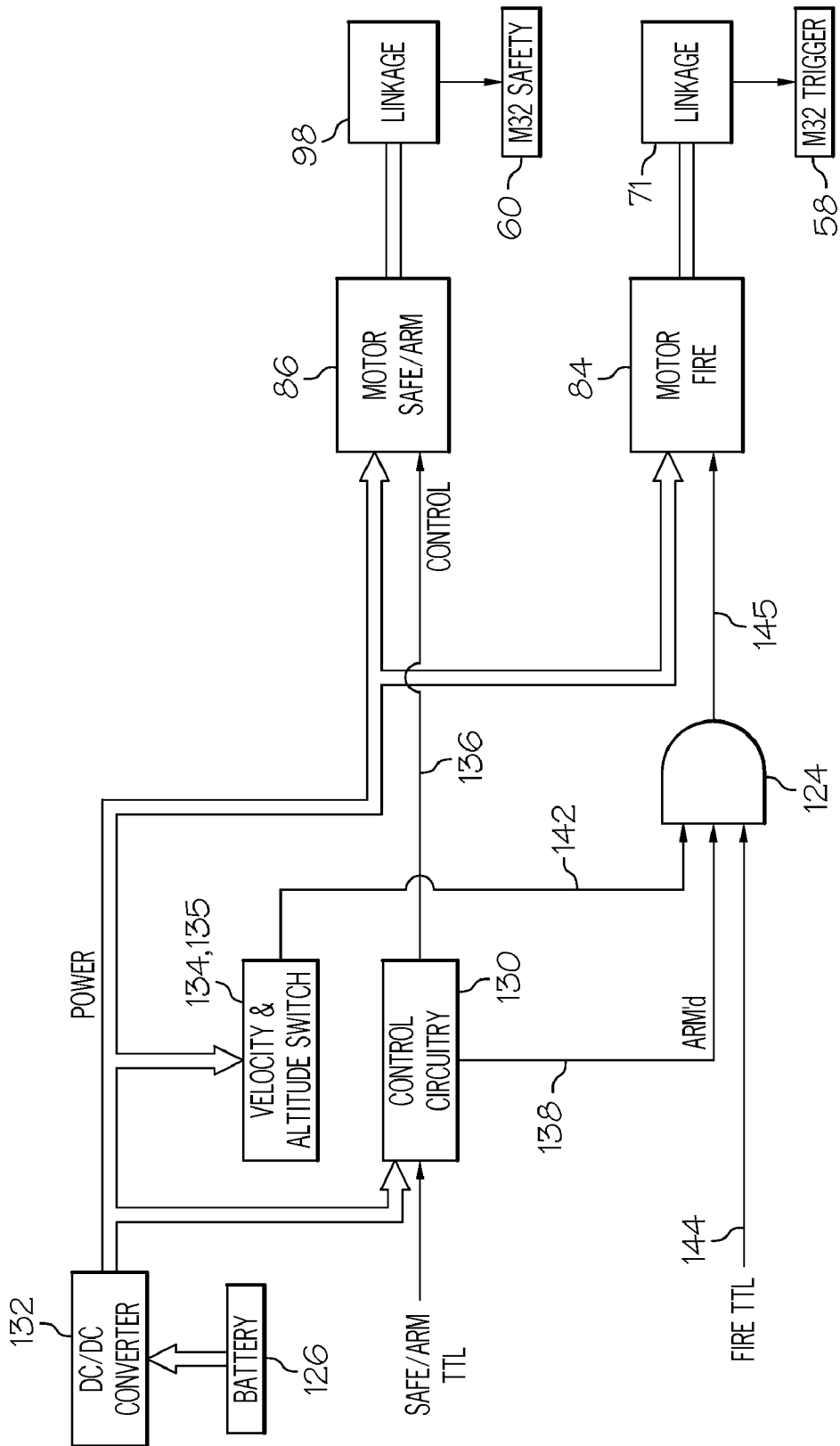


FIG. 17

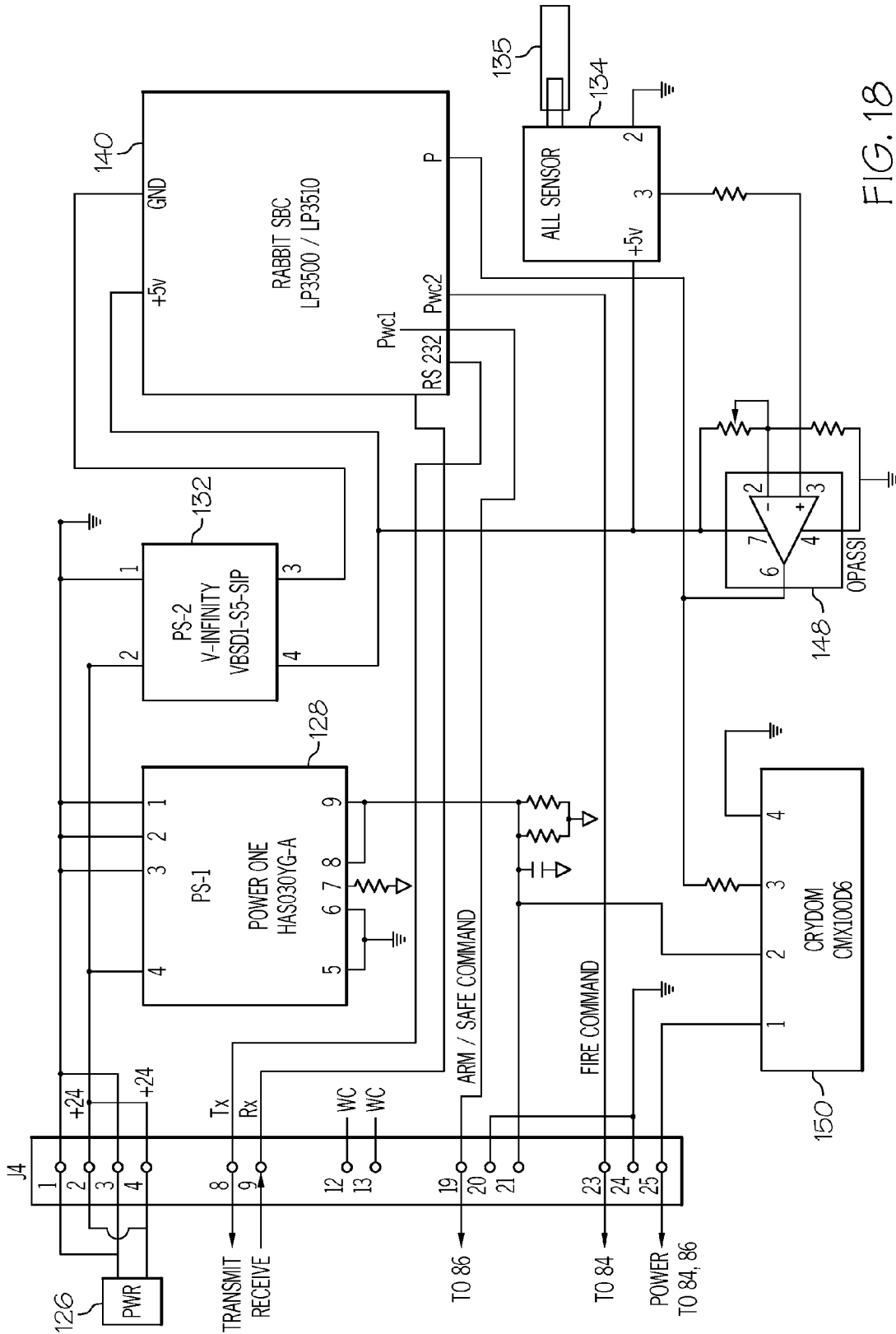


FIG. 18

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## UNMANNED AIR VEHICLE WEAPON ADAPTER

### FIELD OF THE INVENTION

This invention is directed to a weapon adapter for an unmanned aerial vehicle (UAV). More specifically, this invention relates to a system for mounting a weapon, such as an existing handheld weapon, to a UAV and operating the weapon remotely.

### BACKGROUND OF THE INVENTION

Unmanned aerial vehicles are generally known in the art, and vary greatly in size and capability. Larger UAVs are capable of carrying relatively large weapons, such as Hellfire missiles, but tend to be louder, more expensive and easier for an enemy to detect than small, lightweight UAVs. The large weapons typically carry a dedicated rocket motor, and are able to release from the large UAV without imparting considerable force to the UAV.

Smaller UAVs have not been capable of carrying the large weapons commonly associated with aircraft. There remains a need for smaller, lightweight UAVs that are capable of attacking a target.

All US patents and applications and all other published documents mentioned anywhere in this application are incorporated herein by reference in their entirety.

Without limiting the scope of the invention a brief summary of some of the claimed embodiments of the invention is set forth below. Additional details of the summarized embodiments of the invention and/or additional embodiments of the invention may be found in the Detailed Description of the Invention below.

A brief abstract of the technical disclosure in the specification is provided as well only for the purposes of complying with 37 C.F.R. 1.72. The abstract is not intended to be used for interpreting the scope of the claims.

### BRIEF SUMMARY OF THE INVENTION

In some embodiments, an adapter for attaching a handheld firearm to an unmanned air vehicle comprises a mounting assembly configured for attachment to said firearm and to the unmanned air vehicle, and an actuation assembly comprising an actuation device configured to actuate a trigger of the handheld firearm.

In some embodiments, the mounting assembly further comprises a recoil absorbing/isolating device.

In some embodiments, the actuation assembly further comprises a second actuation device configured to actuate a safety mechanism of the handheld firearm.

These and other embodiments which characterize the invention are pointed out with particularity in the claims annexed hereto and forming a part hereof. However, for a better understanding of the invention, its advantages and objectives obtained by its use, reference can be made to the drawings which form a further part hereof and the accompanying descriptive matter, in which there are illustrated and described various embodiments of the invention.

### BRIEF DESCRIPTION OF THE DRAWINGS

A detailed description of the invention is hereafter described with specific reference being made to the drawings.

FIG. 1 shows a side view of a UAV equipped with an embodiment of a handheld weapon and adapter system.

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FIG. 2 shows a top view of a UAV equipped with an embodiment of a handheld weapon and adapter system.

FIG. 3 shows an orthographic view of structural components of a UAV equipped with an embodiment of a handheld weapon and adapter system.

FIG. 4 shows an orthographic view of an embodiment of an adapter system for attaching a handheld weapon.

FIG. 5 shows a top view of an embodiment of an adapter system for attaching a handheld weapon.

FIG. 6 shows a side view of an embodiment of an adapter system for attaching a handheld firearm.

FIG. 7 shows an orthographic view of an embodiment of an adapter system for attaching a handheld firearm.

FIG. 8 shows an exploded view of an embodiment of a recoil isolation mechanism.

FIG. 9 shows an orthographic view of a handheld weapon and an embodiment of an actuation assembly.

FIG. 10 shows an exploded view of an embodiment of an actuation assembly.

FIG. 11 shows an orthographic view of an embodiment of an actuation assembly.

FIG. 12 shows another view of an embodiment of an actuation assembly.

FIG. 13 shows an embodiment of a safety coupling for an embodiment of an actuation assembly.

FIG. 14 shows an exploded view of an embodiment of a safety coupling for an embodiment of an actuation assembly.

FIGS. 15 and 16 show views of an embodiment of a mounting assembly.

FIG. 17 shows an embodiment of a logic diagram for a control and actuation assembly.

FIG. 18 shows an electrical schematic for an embodiment of a control system for the weapon.

### DETAILED DESCRIPTION OF THE INVENTION

While this invention may be embodied in many forms, there are described in detail herein specific embodiments of the invention. This description is an exemplification of the principles of the invention and is not intended to limit the invention to the particular embodiments illustrated.

For the purposes of this disclosure, like reference numerals in the Figures shall refer to like features unless otherwise indicated.

FIG. 1 shows an embodiment of an unmanned aerial vehicle 10 (UAV) comprising an adapter 20 for attaching a weapon 16 to the UAV 10. In some embodiments, the UAV 10 comprises a Killer Bee from Swift Engineering (<http://www.killerbeeuas.com>). In some embodiments, the UAV 10 comprises a wingspan of 10 feet or less, and an unloaded weight of 123 pounds or less. In various embodiments, the weapon 16 is any weapon suitable for use with the UAV 10, which is generally a weapon that can be fired by hand. For example, in the case of a small, relatively lightweight UAV, an existing handheld weapon, such as a pistol, rifle, shotgun, grenade launcher or the like, can be used. In various embodiments, a weapon can be single shot, multiple shot, semi-automatic, fully automatic, etc. As illustrated in FIG. 1, the weapon 16 comprises a grenade launcher such as the M32 Multiple Grenade Launcher or the extended range XRGL40. In some embodiments, the weapon 16 comprises a weight of approximately 14.2 pounds, and the ammunition comprises a rounds at 0.5 pounds each. The total weight of the loaded weapon comprises approximately 17.2 pounds. In some embodiments, the total weight of the UAV 10 with a loaded weapon 16 and the weapon adapter 20 is approximately 150 pounds or less. In some embodiments, recoil forces generated



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by the weapon 16 comprise impulse forces that are considerably higher than the weight of the UAV 10. For example, an M32 grenade launcher can generate impulse forces of 624 pounds or more—more than four times the total weight of the loaded UAV 10 assembly. Thus, in some embodiments, nominal recoil forces acting upon the UAV 10 are greater than 4 Gs. The adapter 20 allows the UAV 10 to maintain stable flight while carrying a weapon 16 that generates G loads of 2, 3, 4 or 5 or more Gs of recoil force.

The weapon 16 defines an axis 17, such as a barrel axis (e.g. the axis along which a launched projectile is initially vectored). Generally, recoil forces caused by the launch of the projectile will act in the direction of the barrel axis 17. The UAV can define a UAV axis 11. In some embodiments, the UAV axis can comprise a central axis of the UAV, and the center of gravity 22 of the UAV is located along the UAV axis. In some embodiments, the weapon/barrel axis 17 intersects the UAV axis 11 at the center of gravity 22. Thus, recoil forces from the weapon 16 are applied through the centroid of the UAV, aiding in stability. In some embodiments, the weight of the adapter 20 is included in determining the center of gravity 22. In some embodiments, the weight of the weapon 16 and the adapter 20 is included in determining the center of gravity 22.

In various embodiments, the weapon axis 17 can have any suitable orientation with respect to the UAV axis. When configured for ground attack, the weapon axis 17 is preferably oriented at an angle  $\alpha$  to the UAV axis 11. The angle  $\alpha$  can be of any suitable magnitude. In some embodiments, the angle  $\alpha$  ranges from 5-20 degrees. In some embodiments, the angle  $\alpha$  is approximately 15 degrees.

FIG. 2 shows a top view of an embodiment of a UAV 10 comprising an adapter 20 for attaching and operating a weapon 16. As shown in the embodiment of FIG. 2, the adapter 20 is centrally stationed in the UAV 10. In some embodiments, an axis 17 of the weapon 16 can be oriented in the same plane 19 as the UAV axis 11. In some embodiments, the plane 19 is vertically oriented. In some embodiments, the adapter 20 comprises a mounting assembly 24 that attaches the weapon 16 to the UAV 10. In some embodiments, the mounting assembly 24 comprises a recoil absorption and/or isolation mechanism 34. In some embodiments, the adapter 20 comprises an actuation assembly 26 arranged to fire the weapon 16.

FIG. 3 shows a partial view of structural elements 18 of an embodiment of a UAV 10, which can include a mounting structure 12 to which the mounting assembly 24 can be attached. In some embodiments, the mounting structure 12 comprises a pair of mounting rails 14. In some embodiments, the rails 14 extend parallel to one another and each defines a flat surface 15 configured for attachment to the mounting assembly 24. The parallel flat surfaces 15 allow for the orientation of the mounting assembly 24 to be easily adjusted, as the mounting assembly 24 can be moved fore and aft, and can also be rotated to adjust the orientation of the axis 17 of the weapon 16 (see FIG. 1). The mounting structure 12 can comprise any suitable configuration for attaching the adapter 20 to the UAV 10. In some embodiments, portions of the UAV mounting structure 12 can extend parallel to the UAV longitudinal axis 11.

FIGS. 4-6 show multiple views of an embodiment of an adapter 20. In some embodiments, the mounting assembly 24 comprises one or more recoil absorbing and/or isolating mechanisms 34, which allows the weapon 16 to move with respect to the UAV 10/12 upon firing. In some embodiments, the mounting assembly 24 comprises a first portion 30 that is moveable with respect to a second portion 32. In some

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embodiments, the first portion 30 is slidably engaged with the second portion 32. In some embodiments, the first portion 30 is attached to the weapon 16 and the second portion 32 is attached to the UAV 10/12. In some embodiments, the first portion 30 is slidably engaged with a pair of second portions 32. Each slidably engaged portion defines an axis of engagement 33 along which the first portion 30 travels with respect to the second portion 32. In some embodiments, each axis of engagement 33 is parallel to the barrel axis 17 of the weapon 16. In some embodiments, each axis of engagement 33 is located an equal distance  $d$  from the barrel axis 17. In some embodiments, the axes of engagement 33 and the barrel axis 17 are oriented in a common plane 35.

In some embodiments, the first portion 30 is configured to suspend the weapon 16 from the second portion 32 such that when the weapon 16 is fired, the recoil mechanism 34 absorbs at least a portion of the recoil forces. The second portion 32 is configured to support the first portion 30 and permit the first portion 30 to move or translate relative to the second portion 32.

In some embodiments, the mounting assembly 24 comprises a pair of recoil mechanisms 34, which are desirably symmetrical and located on opposite sides of the weapon 16.

In some embodiments, the mounting assembly 24 comprises a truss member 36 that supports the handheld firearm 16. In some embodiments, the truss member 36 is rigidly attached to a portion of the weapon 16, such as the barrel. Desirably, the truss member 36 is capable of transmitting moment forces. The truss member 36 can be made from any suitable material and is preferably as lightweight as possible. In some embodiments, the truss member 36 is machined from a piece of billet material, such as aluminum, providing a strong but lightweight component. In some embodiments, the truss member 36 is made of composites, plastics, metals or metal alloys, etc.

Referring to FIG. 7, in some embodiments, the truss member 36 comprises a fixturing portion 54 and a retainer 56. The fixturing portion 54 and retainer 56 are arranged to fixedly clamp the weapon 16 in place. Desirably, the retainer 56 can be removed, allowing the weapon 16 to be removed or exchanged without detaching other portions of the mounting assembly 24.

In some embodiments, the fixturing portion 54 and/or the retainer 56 are configured to engage a retaining mechanism/feature 62 of the weapon 16. In some embodiments, a portion of the weapon 16 can be provided with the retaining feature 62. For example, in some embodiments, a barrel 46 of the weapon comprises a raised hub 62. In some embodiments, the fixturing portion 54 and/or the retainer 56 comprise a cooperative retaining feature, such as a groove, which retains the retaining feature 62 of the weapon 16. In some embodiments, an existing handheld weapon suitable for use as the weapon 16 herein includes the retaining feature 62 as an OEM feature. Thus, the truss member 36 can be specifically designed to engage an existing handheld weapon without requiring modification to the weapon. Various embodiments of the truss member 36 can be designed to engage various types of existing handheld weapons.

In some embodiments, the truss member 36 comprises one or more interchangeable modular adapters. For example, multiple embodiments of a retainer 56 can each be configured to attach to a given truss member 36, wherein each embodiment of the retainer 56 is suitable to engage a different existing weapon.

In some embodiments, a truss member 36 can further comprise a separate adapter (not illustrated) that is configured to engage an embodiment of a fixturing portion 54 and retainer

56. For example, in some embodiments, a fixturing portion 54 and retainer 56 are configured to engage an M32 grenade launcher. The truss member 36 can further comprise an optional separate adapter that allows a smaller caliber weapon to be mounted. For example, a smaller caliber pistol could be mounted, wherein the adapter engages the pistol and the fixturing portion 54 and retainer 56 engage the adapter.

In some embodiments, the retainer 56 is configured to at least partially surround a portion of the weapon 16, such as the barrel 46 and/or the retaining mechanism 62. In some embodiments, the retainer 56 has a plurality of apertures 64, through which fasteners can be placed to engage the fixturing portion 54. Thus, the retainer 56 and the fixturing portion 54 act in conjunction to surround a portion of the weapon 16, such as the barrel 46, thereby fixedly mounting the weapon 16 to the UAV.

In some embodiments, the truss member 36 comprises a lobe or pair of lobes 66, and each lobe 66 is associated with a recoil mechanism 34.

FIG. 8 shows an exploded view of an embodiment of a recoil mechanism 34. A portion of the truss member 36, such as a lobe 66, can be attached to a shaft 42. In some embodiments, the truss member 36 is fixedly attached to the shaft 42. For example, the truss member 36 can be formed with an aperture 68 and a slot 67, configured to clamp the shaft 42. In some embodiments, the aperture 68 and slot 67 are formed by removing material from the truss member 36. The truss member 36 further comprises one or more holes 65 configured to accept a fastener or bolt to tighten the truss member 36 to the shaft 42.

In some embodiments, the shaft 42 is configured to translocate with respect to the second portion 32 of the recoil isolation mechanism 34. In some embodiments, the second portion 32 comprises one or more housing members 44. The housing members 44 are slidably engaged with the shaft 42. In some embodiments, the housing members 44 comprise linear bearings or pillow blocks. The housings 44 support the shaft 42, preferably preventing movement in directions radial to the shaft central axis 43 while allowing movement along the shaft axis 43. In some embodiments, the shaft axis 43 is oriented parallel to the barrel axis 17 of the weapon 16. In some embodiments, the shaft axis is collinear with the axis of engagement 33.

In some embodiments, each housing 44 defines a cavity 45, which can be shaped to receive the shaft 42. Thus, a cross-sectional shape of the cavity 45 is preferably the same as a cross-sectional shape of the shaft 42. In some embodiments, a shaft 42 is circular in cross-section, which can allow some rotational movement between the shaft 42 and the housing 44, for example if the UAV flexes in torsion and the mounting rails 14 move with respect to one another. In some embodiments, the shafts 42 are non-circular in cross-section, for example being polygonal, which can increase structural rigidity of the UAV 10.

In some embodiments, the housing(s) 44 are attached to the UAV structure 12. In some embodiments, the second portion 32 of the recoil mechanism 34 further comprises one or more support brackets 38, wherein each support bracket 38 is attached to the UAV structure 12, and the housing 44 is attached to the support bracket 38. A support bracket 38 allows for the recoil mechanism 34 to be more easily removed from the UAV 10 (e.g. for servicing), and to be easily reinstalled in the proper position.

In some embodiments, the recoil mechanism 34 comprises one or more biasing members 40. The recoil mechanism 34 is configured to damp the shock or recoil that results from firing the weapon 16, and can retard and diminish force transfer

from the weapon 16 to the UAV structure 12. The biasing member 40 is preferably positioned to absorb energy as the first portion 30 of the recoil mechanism 34 translocates from a nominal orientation (e.g. an at rest position, for example as shown in FIG. 7) to a recoiling orientation. Absorbing recoil in the biasing member 40 is desirable because the supporting structure 12 of the UAV 10 is insulated from the sudden shock associated with firing the weapon 16, which aids in stability. The biasing member 40 stores energy until the recoil forces from the firing impulse diminish, for example over a duration of 0.01 seconds or less, and the stored energy is used to move the weapon 16 back to the at rest position. In some embodiments, the biasing member 40 is located between the first portion 30 and the second portion 32 of the recoil mechanism 34. In some embodiments, the biasing member 40 comprises a compression spring. In some embodiments, a spring comprises a helical spring that is wound around a shaft 42.

The first portion 30 of the recoil mechanism 34 can define a recoil side 48 and a rebound side 49. The first portion 30 generally moves in the direction of the recoil side 48 as the weapon 16 and first portion 30 translocate due to initial recoil. The first portion 30 then moves generally in the direction of the rebound side 49 as it rebounds back to the nominal orientation. In some embodiments, a biasing member 40 comprises a compression member and is located to the recoil side 49 of the first portion 30. As shown in FIG. 8, the biasing member 40 is a compression spring located between the rebound side 49 of the first portion 30 and a housing 44 of the second portion 32. In some embodiments, a biasing member 40 comprises a tension member and is located to the rebound side 48 of the first portion 30, for example being attached between the first portion 30 and second portion 32. In some embodiments, a biasing member 40 is capable of functioning as both a tension member and a compression member, for example if the first portion 30 recoils and rebounds back past the nominal orientation.

In some embodiments, the recoil isolation mechanism 34 further comprises a damper 50 or bumper. In some embodiments, the damper 50 is disposed between the first portion 30 and a portion of the second portion 32, such as a housing 44 located opposite the biasing member 40. As illustrated in FIG. 8, the damper 50 is oriented to absorb shock from the first portion 30 as the first portion 30 rebounds back to the nominal orientation and/or past the nominal orientation. In some embodiments, it is preferable to use a damper 50 in addition to the biasing member 40, for example when the biasing member 40 is configured to only function as a compression member. In some embodiments, the biasing member 40 is loaded with a predetermined amount of compression in the at rest orientation, and the damper 50 can experience similar compressive forces. In some embodiments, the damper 50 functions as a secondary biasing member oriented to damp any impact between the first portion 30 and the rebound side 49 housing 44. In some embodiments, the damper 50 comprises a resilient material, such as an elastomeric material. In some embodiments, the initial recoil forces of the weapon 16 can impart a pitch, yaw or roll motion to the UAV, for example when the barrel axis 17 is offset slightly from the UAV center of mass 22 (see FIG. 6). Such motion is generally counteracted when the first portion 30 of the mounting assembly 24 rebounds against the damper 50 as the weapon 16 returns to the at rest orientation.

In various embodiments, the biasing member 40 can comprise any mechanism suitable for functioning as a biasing member/mechanism 40. For example, in some embodiments, an air or gas spring is used, such as a sealed device wherein positive and/or negative gas pressure provides biasing forces.

In some embodiments, a hydraulic or other fluid resistance device is used. In some embodiments, a fluid resistance device comprises a non-Newtonian fluid. In some embodiments, a magnetic resistance device using permanent and/or electromagnets can provide biasing forces.

In alternative embodiments of the recoil mechanism **34**, a shaft **42** comprises the second portion **32** of the recoil mechanism **34**. For example, in some embodiments, the shaft **42** is fixedly attached to a housing **44**, a support bracket **38**, or a portion of the UAV structure **12**. The first portion **30**, such as the truss member **36**, is slidably engaged with the shaft **42**, configured to slide in the direction of the shaft axis **43**.

FIG. 9 shows an embodiment of an actuation assembly **26** attached to a weapon **16**. In some embodiments, the UAV weapon adapter **20** comprises an actuation assembly **26** configured to actuate a trigger **58** of the weapon **16**. In some embodiments, the actuation assembly **26** is affixed to the weapon **16**, and therefore travels with the weapon, for example as the weapon **16** recoils. This allows the actuation assembly **26** to fire the weapon **16** at any time.

In some embodiments, the actuation assembly **26** comprises a body member **80** and a first actuation device **84**. In some embodiments, the actuation assembly **26** is configured to operate a safety mechanism **60** of the weapon **16**, and comprises a second actuation device **86**. In some embodiments, the body member **80** supports the first actuation device **84** and the second actuation device **86**.

In some embodiments, the body member **80** comprises mounting brackets **82**, **83** that allow attachment to the weapon **16**. For example, a first mounting bracket **82** allows engagement with a butt section **76** of the weapon **16**. A second mounting bracket **83** is configured to engage a second mounting section **77** of the weapon **16**. Each mounting bracket **82**, **83** comprises a hole **92**, through which a fastener is preferably disposed to engage the weapon **16**.

In alternative embodiments, the body member **80** can be otherwise configured to engage the weapon **16**, for example using clamps, quick release locks, or the like.

In some embodiments, the weapon **16** is mounted on a first side **87** of the body member **80**, and is secured to the body member **80** by a first mounting bracket **82**. The first and second actuation devices **84**, **86** are mounted to a second side **88** of the body member **80**, and are configured to extend through the body member **80** to engage the respective trigger **58** and safety mechanism **60** of the weapon **16**.

FIG. 10 shows the weapon **16** disengaged from the body member **80**, and an exploded view of an embodiment of a trigger actuation mechanism **71**, which links the first actuation device **84** to the trigger **58**.

In some embodiments, the first and second actuation devices **84**, **86** each comprise a servo motor. In some embodiments, a servo motor comprises a lightweight, high-torque servo motor such as a Futaba S3306 1/5 scale motor available from Futaba Radio Control Systems. In some embodiments, the first and second actuation devices **84**, **86** comprise the same type of motor. In some embodiments, the first and second actuation devices **84**, **86** comprise different types of motors, for example being individually selected according to the torque necessary to operate the trigger **58** or the safety **60**.

In some embodiments, the first and second actuation devices **84**, **86** are attached to the body member **80** via with fasteners **89**, such as complimentary bolts and tube nuts. In some embodiments, the actuation devices **84**, **86** are prefabricated and are provided with mounting tabs **93**. In some embodiments, a spacer **90**, such as a stand-off tube, is used

with a fastener **89**. A spacer **90** allows the fasteners **89** to be properly tightened without risk of damaging or breaking off the mounting tabs **93**.

The trigger actuation mechanism **71** is oriented to transmit force from the first actuation device **84** to the trigger **58**. The trigger actuation mechanism **71** desirably moves in accordance with the movement of the trigger **58**. In some embodiments, the trigger actuation mechanism **71** comprises an actuation lever **74** that engages the first actuation device **84**. For example, the actuation lever **74** can engage an output shaft **85** of a servo motor. The trigger actuation mechanism **71** further comprises a trigger engaging portion **72** configured to actuate the trigger **58**. In some embodiments, the trigger engaging portion **72** comprises a roller, which is preferable when the trigger **58** comprises curvature. A roller is desirably capable of rolling along the trigger **58** as the trigger **58** and the actuation lever **74** move during actuation. In some embodiments, the trigger engaging portion **72** is rotatably engaged with the actuation lever **74** or a shaft **78** attached to the actuation lever **74**. In some embodiments, a shaft **78** can be secured to the actuation lever **74** with fasteners, such as a plurality of nuts and washers.

In some embodiments, the second actuation device **86** is coupled to the safety mechanism **60** of the weapon **16**. Thus, the second actuation device **86** is capable of operating the safety **60**, including both engaging the safety **60** to prevent fire and defeating the safety **60** to allow fire. In some embodiments, an output portion **91** or shaft of the second actuation device **86** is shaped/configured to engage an existing portion of a safety mechanism **60** of an OEM weapon, without modification to the safety mechanism **60**.

In various embodiments, the actuation devices **84**, **86** comprise any actuation mechanism suitable for accomplishing its task as described herein. For example, any suitable rotational motor can be used, such as a stepping motor, servo motor, and the like. Further, in some embodiments, a linear actuation mechanism can be used for either actuation device **84**, **86**. Various actuation devices **84**, **86** can be selected according to the specifics of the weapon **16** being used. In some embodiments, the first actuation device **84** comprises a different type of actuation mechanism than the second actuation device **86**. For example, in some embodiments, the first actuation device **84** comprises a linear actuator and the second actuation device **86** comprises a rotational actuator.

FIG. 11 shows another embodiment of an actuation assembly **26**. In some embodiments, the actuation assembly **26** comprises a releasable link coupling **98** between the second actuation device **68** and the safety mechanism **60** of the weapon **16**. The releasable link coupling **98** allows for manual operation of the safety mechanism **60** without use of the second actuation device **68**, for example by a service technician, without any risk of damage to the second actuation device **68**. An embodiment of a releasable link coupling **98** is discussed in detail below.

In some embodiments, an actuation assembly **26** comprises a safety locking mechanism **99** configured to lock the safety mechanism **60** of the weapon **16** in a safe/no-fire orientation. For example, a service technician may engage the safety locking mechanism **99** during servicing of the weapon **16** to guarantee that the weapon **16** will not fire. In some embodiments, the body member **80** of the actuation assembly **26** comprises an extension portion **81**. In some embodiments, the safety locking mechanism **99** is mounted to the extension portion **81**, and extends through the extension portion **81**.

FIG. 12 shows an embodiment of an actuation assembly **26** engaged with a weapon **16**. The safety **60** of the weapon **16** is

coupled to the second actuation device **86** via an embodiment of a releasable link coupling **98**.

In some embodiments, a safety locking mechanism **99** is moveable between first and second positions. In a first position, the safety locking mechanism **99** does not interfere with operation of the safety mechanism **60**. Thus, the safety mechanism **60** can be engaged and/or disengaged, for example manually or by the second actuation device **86**. In the second position, the safety locking mechanism **99** is positioned to interfere with the safety mechanism **60**, locking the safety mechanism **60** in a safe/no-fire orientation. If the second actuation device **86** actuates to move the safety **60** to a fire orientation when the safety locking mechanism **99** is in the second/interference position, the releasable link coupling **98** will disengage, and the safety **60** will not be de-activated. In some embodiments, the safety locking mechanism **99** comprises detents that encourage the safety locking mechanism **99** to assume the first or second position.

FIG. **13** shows an embodiment of a releasable link coupling **98**. FIG. **14** shows an exploded view of an embodiment of the releasable link coupling **98**. In some embodiments, the releasable link coupling **98** comprises a first portion **41** releasably couples with a second portion **47**. The first portion **41** is attached to an output portion/shaft the second actuation device **86**, and the second portion **47** is attached to the safety mechanism **60** of the weapon **16**. The releasable link coupling **98** is preferably capable of transmitting a predetermined amount of torque between the first portion **41** and second portion **47**, and will release if an more than the predetermined amount is attempted to be transferred. For example, the releasable link coupling **98** will release if the safety mechanism **60** is manually operated without actuation of the second actuation device **86**, and/or if the second actuation device **86** is actuated when the safety mechanism **60** is unable to move, for example due to a safety locking mechanism **99**.

In some embodiments, the first portion **41** of the releasable link coupling **98** comprises a spindle **102** that is engaged to the second actuation device **86**. In some embodiments, a spindle **102** comprises a second actuation output **91**. In some embodiments, an adapter **96** is used between the second actuation device **86** and the spindle **102**. In some embodiments, an adapter **96** can be used to attach the spindle **102** to an output of a servo motor, for example using fasteners **114** or any other suitable attachment method. In some embodiments, a sleeve **97** is oriented about a portion of the spindle **102**. A sleeve **97** can be used to center the spindle **102** as it passes through an aperture in the body member **80** of the actuation assembly **26**. In some embodiments, a sleeve **97** comprises a bushing. In some embodiments, the sleeve **97** is stationary relative to the body member **80**, while the spindle **102** rotates relative to the sleeve **97**. In some embodiments, the sleeve **97** is friction fit on the spindle **102** and rotates with the spindle **102**. In some embodiments, a sleeve **97** is formed from a low friction material. In some embodiments, a sleeve **97** functions as a vibration damping/insulating member.

In some embodiments, the spindle **102** comprises a plurality of slots **108** or grooves, which extend into an outer surface of the spindle **102**. As shown in FIG. **14**, the spindle **102** comprises three (3) slots **108**, which are evenly distributed about the outer circumference of the spindle **102**. In some embodiments, the slots **108** collectively comprise a detent that the second portion **47** of the releasable link coupling **98** engages.

In some embodiments, the second portion **47** of the releasable link coupling **98** comprises one or more engaging members **112** arranged to engage the first portion **41**. In some embodiments, an engaging member **112** comprises a ball

locking mechanism **111**. For example, as shown in FIG. **14**, each ball locking mechanism **111** engages a slot **108** in the spindle **102**. In some embodiments, ball locking mechanism **111** comprises a threaded body and a ball bearing **113** disposed at an end of the ball locking mechanism **111**. A spring (not illustrated) located within the threaded body biases the ball **113** in an outward direction, and the ball **113** can move into the threaded body upon the application of an inward force. Each ball locking mechanism **111** is threaded into a coupler body **94** of the second portion **47**. Thus, the ball locking mechanisms **111** shown in FIG. **14** each engage a slot **108** in the spindle, thereby engaging the first portion **41** with the second portion **47**. If the first portion **41** and second portion **47** are turned against one another with enough force to overcome the ball locking mechanisms **111**, the balls **113** move out of the slots **108**, decoupling the first portion **41** from the second portion **47**. As the slots **108** are rotated back into alignment with the ball locking mechanisms **111**, the first portion **41** and second portion **47** reengage. In some embodiments, a slot **108** comprises an arcuate or semi-circular cross-sectional shape. In some embodiments, a cross-sectional shape of the slot **108** is shaped to cooperatively engage a portion of a ball locking mechanism **111**.

In some embodiments, the coupler body **94** comprises a shaped body portion that provides an aperture for each engaging member **112**. In some embodiments, the releasable link coupling **98** further comprises a link member **101** attached between the coupler body **94** and the safety mechanism **60** of the weapon **16**. The link member **101** transmits the rotational motion of the coupler body **94** to the safety mechanism **60**. In some embodiments, the link member **101** comprises a shaped aperture **120** that engages a shaped protrusion **122** on the safety **60**. The engagement between the shaped aperture **120** and the shaped protrusion **122** preferably transmits a greater amount of torque than can be transmitted between the first portion **41** and second portion **47** of the releasable link coupling **98**. In some embodiments, the shaped aperture **120** is shaped to engage a protrusion **122** that exists in the safety mechanism **60** of an OEM weapon **16** without modification.

In some embodiments, the coupler body **94** further comprises or is arranged to engage a transfer pin **106**. The transfer pin **106** is offset from a central axis of the spindle **102**. In some embodiments, the transfer pin **106** engages the link member **101**, thereby transmitting torque to the link member **101** and safety mechanism **60**.

FIG. **13** also shows an embodiment of a safety locking mechanism **99** oriented in an interference position, wherein the safety mechanism **60** is locked in a safe orientation. As shown in FIG. **14**, a detent mechanism can be used that encourages the safety locking mechanism **99** to be positioned in either the first position or the second position. For example, a ball locking mechanism **111** can be used to engage the safety locking mechanism **99**. In some embodiments, a ball locking mechanism **111** can be received in the extension portion **81** (see FIG. **11**) of the body member **80**, positioned such that the ball **113** engages a detent slot, groove or depression in the safety locking mechanism **99**.

FIGS. **15** and **16** illustrate different views of an embodiment of a truss member **36**. The truss member **36** preferably comprises a three-dimensional truss capable of transmitting tension, compression and moment forces between the weapon support/fixturing portion **54** and the outer portions of the lobes **66**, which transmit forces to other portions of the mounting assembly **24** and/or the UAV. In some embodiments, the truss member **36** comprises opposed wall portions **37**, **39** and a plurality of cross-linking members **51**. Various cross-linking members can be oriented orthogonally with

respect to one another. In some embodiments, the truss member **36** comprises a plurality of apertures **61**, for example to save weight.

In some embodiments, the UAV is controlled remotely. Thus, an operator inputs flight and weapon commands into a control box, which is generally located on the ground. Flight and weapon commands are transmitted wirelessly to the UAV.

FIG. 17 shows an embodiment of a logic flow diagram and components located in the UAV, which interpret and accomplish the commands received from the ground control box. A power source **126** supplies power to control circuitry **130**, actuation mechanisms **84**, **86** and any other equipment that requires power. In some embodiments, a power converter **132** is used, for example when the circuitry and/or actuation mechanisms **84**, **86** require a different voltage than is provided by the power source **126**. The control circuitry **130** receives commands pertinent to the safety mechanism **60** of the weapon **16**. The control circuitry **130** outputs a safety command **136** to the second actuation device **86**, for example engaging or disengaging the safety mechanism **60**. When the safety mechanism **60** is disabled and the weapon can fire, the control circuitry **130** outputs an armed signal to an arming mechanism **124**. In some embodiments, an altitude sensor **134** and/or a velocity sensor **135** are provided as a safety mechanism, which guarantee that the UAV has reached a predetermined altitude and/or airspeed prior to allowing the weapon to fire. When the UAV is above the predetermined altitude and/or airspeed, the altitude sensor **134** and/or velocity sensor **135** output an appropriate signal **142** to the arming mechanism **124**. When a fire command is received from the ground control box, a fire signal **144** is transmitted to the arming mechanism **124**. If the appropriate armed signal **138** and altitude/velocity signal **142** are also received, the arming mechanism **124** outputs a fire command **145** to the first actuation mechanism **84**, thereby firing the weapon.

FIG. 18 shows an electrical schematic for an embodiment of an actuation assembly **26**. In some embodiments, the parts shown in the schematic are included on a circuit board attached to the body member **80** of the actuation assembly **26**. In some embodiments, a circuit board is attached to another portion of the UAV, and appropriate connections are made between the illustrated parts.

FIG. 18 shows a terminal assembly **J4**, which in some embodiments comprises the main connector for the circuit board. Pins **1** and **3** comprise ground connections and pins **2** and **4** comprise positive power connections, for example being attached to a power source **126**. In some embodiments, the circuit comprises a power converter **128**, such as a DC-DC converter that converts the voltage received from the source **126** to a level suitable for other components of the circuit. For example, in some embodiments, a circuit computer **140**, safety sensors **134**, **135** and actuation devices **84**, **86** receive a down-converted 5 volt supply, whereas other electronics in the UAV receive 24 volts from the source **126**. In some embodiments, the power converter **128** comprises a power conditioner, which provides a clean supply of power to the circuit and isolates the circuit from other components that receive power from a common power supply. In some embodiments, the power converter **128** comprises a model HAS030YG-A HAS series DC-DC converter, available from Power One (<http://www.power-one.com/>).

In some embodiments, the circuit comprises a second power converter **132**, which regulates and isolates power delivered to certain portions of the circuit, such as the computer **140**, from other portions of the circuit, such as output to the actuation devices **83**, **86**. In some embodiments, the sec-

ond power converter **132** comprises a model VBSD1-S5-S5-SIP available from V-Infinity, 10050 SW 112<sup>th</sup> Ave., Tualatin, Oreg. 97062 ([www.cui.com](http://www.cui.com)).

Pin **9** of the terminal assembly **J4** comprises a receive channel Rx that receives commands from the ground control box, such as safety mechanism engaging/disengaging commands and a weapon fire command, and sends the commands to the computer **140**. Pin **8** of the terminal assembly **J4** comprises a transmit channel Tx, which relays status information from the computer **140** back to the ground control box. In various embodiments, status information includes the status of any and all mechanisms in the circuit. In some embodiments, the receive channel Rx and transmit channel Tx are carried over an RS232 connection.

The computer **140** comprises a first pulse waveform output channel PWC1, which communicates with the second actuation mechanism **86** over pin **19** of the terminal assembly **J4**. Thus, the first pulse waveform output channel PWC1 controls the second actuation mechanism **86** and therefore the safety mechanism **60** of the weapon **16**. When the computer **140** receives a safety mechanism **60** engage/disengage command via the receive channel Rx, it outputs an appropriate pulse waveform command via PWC1. In some embodiments, the computer **140** comprises a model LP3500 Low-Power Single-Board Computer microprocessor available from Rabbit Semiconductor, 2900 Spafford Street, Davis, Calif. 95618 (<http://www.rabbit.com>).

In some embodiments, the circuit comprises various safety circuits that must be satisfied prior to providing power to the first actuation device **84**, or to both actuation devices **84**, **86**. In some embodiments, the safety circuits include an altitude sensor **134**, which only allows weapon firing over a predetermined altitude. In some embodiments, the safety circuits include an airspeed sensor **135**, such as a positive pressure device that uses pressure provided by forward motion to output a variable voltage. In some embodiments, an output of the airspeed sensor **135** is provided to an operational amplifier **148**, which provides a voltage to a relay **150** only after an appropriate voltage corresponding to a predetermined minimum airspeed is received. When the relay **150** receives the voltage/arming signal from the op amp **148**, it provides power to the first actuation device **84**, or to both actuation devices **84**, **86** via pin **25** of the terminal assembly **J4**. In some embodiments, the relay **150** comprises a model CMX100D6 solid-state MOSFET relay available from Crydom, Inc., 2320 Paseo de las Americas, Suite 201 San Diego, Calif. 92154 (<http://www.crydom.com>).

In some embodiments, the output voltage of the op amp **148** is also provided to the computer **140**, which verifies that the safety circuitry requirements have been satisfied prior to sending a weapon fire signal. The computer **140** comprises a second pulse waveform output channel PWC2, which communicates with the first actuation mechanism **84** over pin **23** of the terminal assembly **J4**. Thus, the second pulse waveform output channel PWC2 controls the first actuation mechanism **84** and therefore the trigger **58** of the weapon **16**. When the computer **140** receives a weapon fire command via the receive channel Rx, and a safety circuit signal from the op amp **148**, it outputs an appropriate pulse waveform command via PWC2.

The above disclosure is intended to be illustrative and not exhaustive. This description will suggest many variations and alternatives to one of ordinary skill in this field of art. All these alternatives and variations are intended to be included within the scope of the claims where the term "comprising" means "including, but not limited to." Those familiar with the art

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may recognize other equivalents to the specific embodiments described herein which equivalents are also intended to be encompassed by the claims.

Further, the particular features presented in the dependent claims can be combined with each other in other manners within the scope of the invention such that the invention should be recognized as also specifically directed to other embodiments having any other possible combination of the features of the dependent claims. For instance, for purposes of claim publication, any dependent claim which follows should be taken as alternatively written in a multiple dependent form from all prior claims which possess all antecedents referenced in such dependent claim if such multiple dependent format is an accepted format within the jurisdiction (e.g. each claim depending directly from claim 1 should be alternatively taken as depending from all previous claims). In jurisdictions where multiple dependent claim formats are restricted, the following dependent claims should each be also taken as alternatively written in each singly dependent claim format which creates a dependency from a prior antecedent-possessing claim other than the specific claim listed in such dependent claim below.

This completes the description of the preferred and alternate embodiments of the invention. Those skilled in the art may recognize other equivalents to the specific embodiment described herein which equivalents are intended to be encompassed by the claims attached hereto.

What is claimed is:

1. An adapter for attaching a handheld firearm to an unmanned air vehicle comprising:
  - a mounting assembly configured for attachment to said firearm and to said unmanned air vehicle; and
  - an actuation assembly comprising a first actuation device configured to actuate a trigger of said handheld firearm, a second actuation device configured to actuate a safety mechanism of said handheld firearm and a releasable link coupling between the second actuation device and the safety mechanism, said releasable link coupling arranged to release an attachment between said second actuation device and said safety mechanism upon application of a predetermined amount of torque.
2. The adapter of claim 1, wherein said mounting assembly comprises a recoil isolation mechanism.
3. The adapter of claim 1, wherein said mounting assembly comprises a first portion slidably engaged with a second portion.
4. The adapter of claim 3, wherein said mounting assembly further comprises a biasing member.
5. The adapter of claim 4, wherein said biasing member is positioned between the first portion and the second portion.
6. The adapter of claim 5, further comprising a damper, wherein said biasing member and said damper are located on opposite sides of the first portion.
7. The adapter of claim 4, wherein said biasing member forces said first portion to a nominal position.
8. The adapter of claim 1, wherein said mounting assembly comprises a truss member fixedly attached to said handheld firearm.
9. The adapter of claim 8, wherein said mounting assembly further comprises a support bracket, said truss member slidably engaged with said support bracket.

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10. The adapter of claim 9, further comprising a shaft, said shaft being slidably engaged with said support bracket or said truss member.

11. The adapter of claim 10, wherein a longitudinal axis of said shaft is parallel to a longitudinal axis of a barrel of said handheld firearm.

12. The adapter of claim 1, wherein said actuation assembly comprises a body configured to be supported by a body of said handheld firearm.

13. The adapter of claim 1, said actuation assembly further comprising a locking mechanism configured to prevent actuation of said safety mechanism.

14. The adapter of claim 1, wherein said actuation device comprises a servo motor.

15. The adapter of claim 1, wherein a barrel axis of the handheld firearm passes through a center of gravity of the UAV.

16. The adapter of claim 1, wherein said actuation assembly comprises a roller configured to contact said trigger.

17. An unmanned air vehicle comprising:

- a structural framework;
- a commercially available firearm designed to be fired by hand, the firearm comprising a trigger and a safety;
- a mounting assembly, a first portion of said mounting assembly attached to said firearm, a second portion of said mounting assembly attached to said structural framework;
- a first actuator configured to actuate said trigger;
- a second actuator configured to actuate said safety; and
- a releasable link coupling between the second actuation device and the safety mechanism, said releasable link coupling arranged to release an attachment between said second actuation device and said safety mechanism upon application of a predetermined amount of torque.

18. The unmanned air vehicle of claim 17, wherein said mounting assembly comprises a recoil absorbing mechanism that isolates the structural framework from recoil forces generated by the firearm.

19. The unmanned air vehicle of claim 18, wherein said mounting assembly only contacts a barrel of the firearm.

20. An unmanned air vehicle comprising:

- a structural framework;
- a firearm designed to be fired by hand, the firearm comprising a body, a trigger, a safety and a barrel;
- a mounting assembly attaching said firearm to said structural framework, a first portion of said mounting assembly attached to said barrel of said firearm, a second portion of said mounting assembly attached to said structural framework, said mounting assembly comprising a recoil absorbing mechanism; and
- a first actuator configured to actuate said trigger; a second actuator configured to actuate said safety; and a releasable link coupling between the second actuation device and the safety mechanism, said releasable link coupling arranged to release an attachment between said second actuation device and said safety mechanism upon application of a predetermined amount of torque; wherein said mounting assembly only contacts the barrel of the firearm.

21. The unmanned air vehicle of claim 20, wherein said firearm is a grenade launcher.

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